

NOZZLE Program

Version 4.0

User Guide

April 2016

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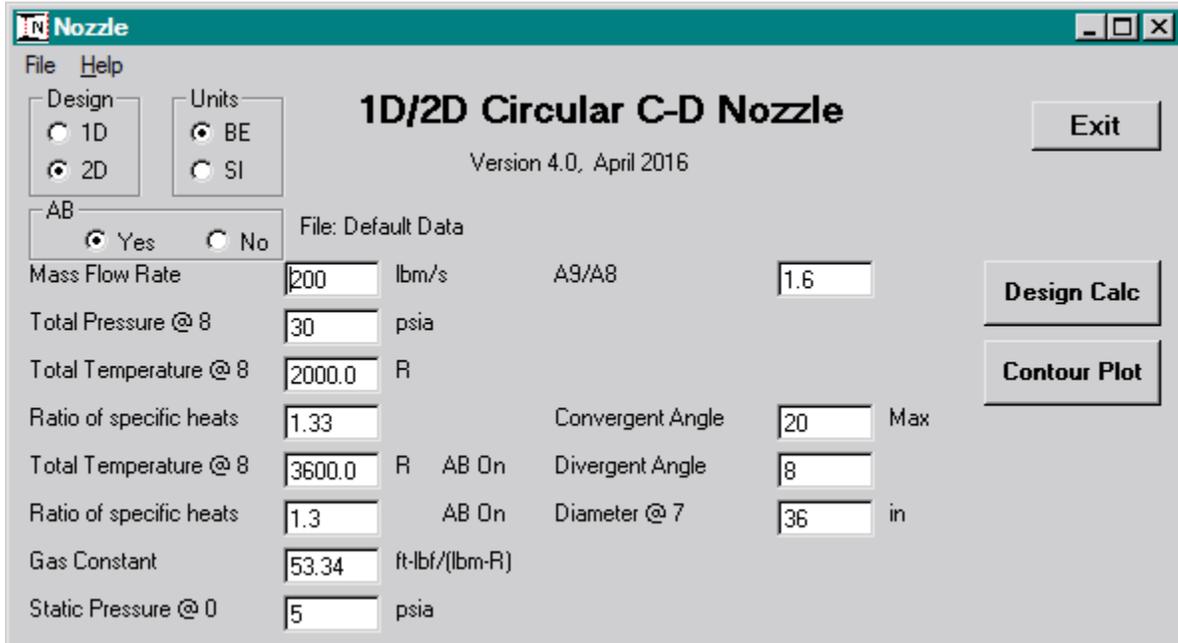
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1. MAIN Window

The NOZZLE conceptual design program analyzes both 1-D and 2-D circular convergent-divergent nozzles. It is based on the equations developed in Chapter 10 of *Aircraft Engine Design, Second Edition*.

This program is designed to be user-friendly and multiple windows are used for program control and data input. NOZZLE input data files may be saved on disk for later use (the file extension “NOZ” is used for these files). Also, saved input data files may be read from disk for current use. The current data file name is displayed in the upper left of the Main window and in output windows. Program output is directed to output windows and may be sent to a printer.

When the NOZZLE program is run, the Main window is displayed. The top of the Main window is shown below.



2. Design Calculation

Pressing the **Design Calc** button causes the program to calculate the nozzle performance using the input design data and displaying the results in the output window. The one-dimensional (1-D) analysis has only dimensions normal to the flow direction (flow areas) and does not have length dimensions. The results for a 1-D circular nozzle are shown below.

The screenshot shows the 'Nozzle' software interface. The title bar reads 'Nozzle'. The main window title is '1D/2D Circular C-D Nozzle' with the version 'Version 4.0, April 2016'. The interface includes a menu bar with 'File' and 'Help'. There are two groups of radio buttons: 'Design' with '1D' selected and '2D' unselected; 'Units' with 'BE' selected and 'SI' unselected. Below these are 'AB' radio buttons with 'Yes' selected and 'No' unselected, and a 'File: Default Data' label. The input section contains several fields: 'Mass Flow Rate' (200 lbm/s), 'Total Pressure @ 8' (30 psia), 'Total Temperature @ 8' (2000.0 R), 'Ratio of specific heats' (1.33), 'Total Temperature @ 8' (3600.0 R), 'Ratio of specific heats' (1.3), 'Gas Constant' (53.34 ft-lbf/(lbm-R)), and 'Static Pressure @ 0' (5 psia). There are also fields for 'A9/A8' (1.6), 'Pi N (Pt9/Pt7)' (0.98), and 'Discharge Coeff (CD)' (0.98). On the right side, there are buttons for 'Exit', 'Design Calc', 'Plot', and 'Print'. At the bottom, a text area titled 'File: Default Data' displays the following calculated results:

```
File: Default Data

m dot = 200.0 lbm/sec   Pi N = 0.9800
Mach 8 = 1.000         Area 8 = 4.044 ft^2
Mach 9 = 1.902         Area 9 = 6.471 ft^2
P 9 = 4.460 psia      V 9 = 3215.4 ft/sec
Mach 9i = 1.925       V 9i = 3239.9 ft/sec
P 9i = 4.386 psia     V s = 3151.1 ft/sec
V s = 3151.1 ft/sec   Fg a = 19484 lbf
Fg a = 19484 lbf      Fg i = 19988 lbf
CV = 0.9924           CD = 0.9800
Cf g = 0.9748         P9/P0 = 0.8920
```

The two-dimensional (2-D) analysis has both dimensions normal to the flow direction (flow areas) and length dimensions. The results for an example 2-D circular convergent-divergent nozzle are shown below. Note that the nozzle can be sketched for the 2-D analysis.

The screenshot shows the 'Nozzle' software interface. The title bar reads 'Nozzle'. The main window title is '1D/2D Circular C-D Nozzle', with 'Version 4.0, April 2016' below it. The interface includes a menu bar with 'File' and 'Help'. There are several control panels: 'Design' with radio buttons for '1D' and '2D' (2D is selected); 'Units' with radio buttons for 'BE' and 'SI' (BE is selected); and 'AB' with radio buttons for 'Yes' and 'No' (Yes is selected). A 'File: Default Data' label is present. The main input area contains the following fields:

- Mass Flow Rate: 200 lbm/s
- Total Pressure @ 8: 30 psia
- Total Temperature @ 8: 2000.0 R
- Ratio of specific heats: 1.33
- Total Temperature @ 8: 3600.0 R
- Ratio of specific heats: 1.3
- Gas Constant: 53.34 ft-lbf/(lbm-R)
- Static Pressure @ 0: 5 psia
- A9/A8: 1.6
- Convergent Angle: 20 Max
- Divergent Angle: 8
- Diameter @ 7: 36 in

On the right side, there are buttons for 'Exit', 'Design Calc', 'Contour Plot', 'Sketch', and 'Print'. Below the input fields is a text area titled 'File: Default Data' containing the following calculated results:

```

File: Default Data

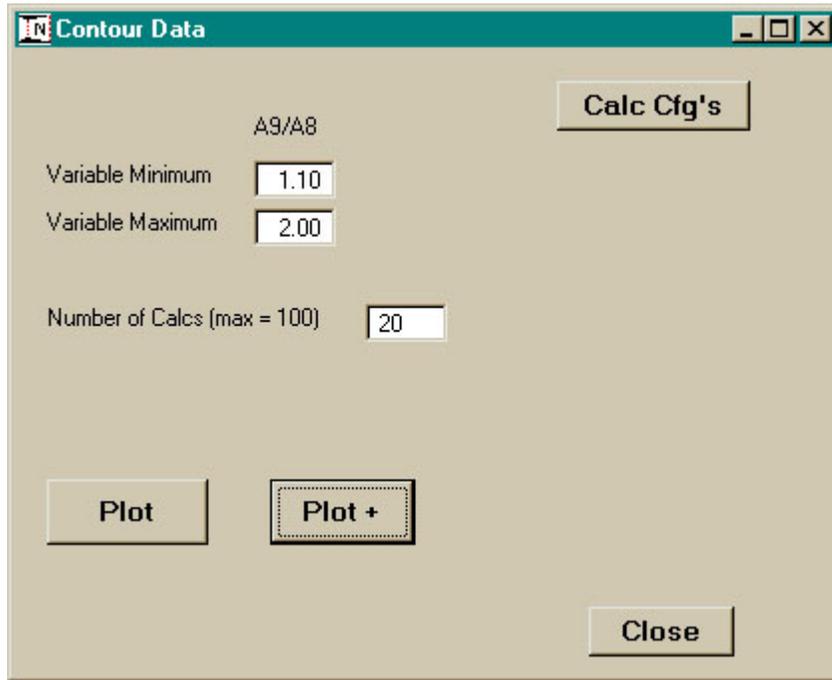
m dot = 200.0 lbm/sec
Mach 7 = 0.351
Mach 8 = 1.000
Mach 9 = 1.938
P 9 = 4.205 psia
Mach 9i = 1.951
P 9i = 4.205 psia
V s = 3151.1 ft/sec
Fg a = 19318 lbf
CD = 0.9575
CV = 0.9957
Pi N = 0.9882
Cfg = 0.9443

Area 7 = 7.069 ft^2
Area 8 = 4.139 ft^2
Area 9 = 6.623 ft^2
V 9 = 3253.4 ft/sec
V 9i = 3267.3 ft/sec
Fg i = 20457 lbf
Cfgpeak = 0.9885
CA = 0.9927
P9/P0 = 0.8411

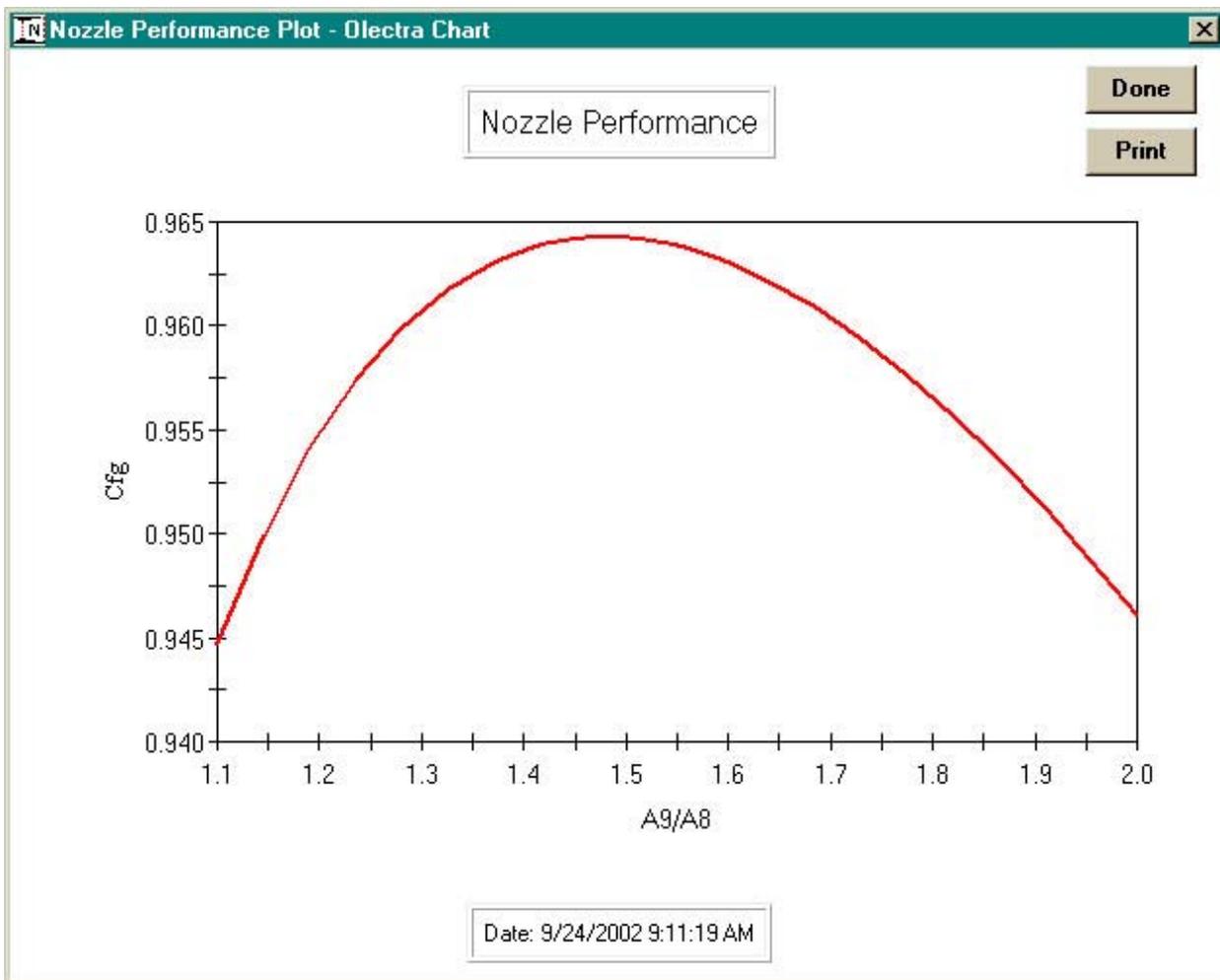
```

3. Gross Thrust Coefficient Contours

The variation of the gross thrust coefficient (C_{fg}) with design variable(s) can be calculated and presented using the **Contours/Plot** button. The following data window is displayed for the 1-D circular nozzle. The **Plot** and **Plot+** buttons become active after the **Calc Cfg's** button has been pressed and the performance calculated. Similar to the INLET program, the **Plot** button displays a plot where the mouse arrow becomes a cross-hair when the user moves the mouse across the plot. This feature allows the user to capture and display at the bottom of the screen the coordinates and total pressure value by pressing the left mouse button. Pressing the **Plot+** button produces an Olectra Chart plot of the gross thrust coefficient versus nozzle area ratio.



The plot generated by the **Plot+** button for the 1-D circular convergent-divergent nozzle is shown below.



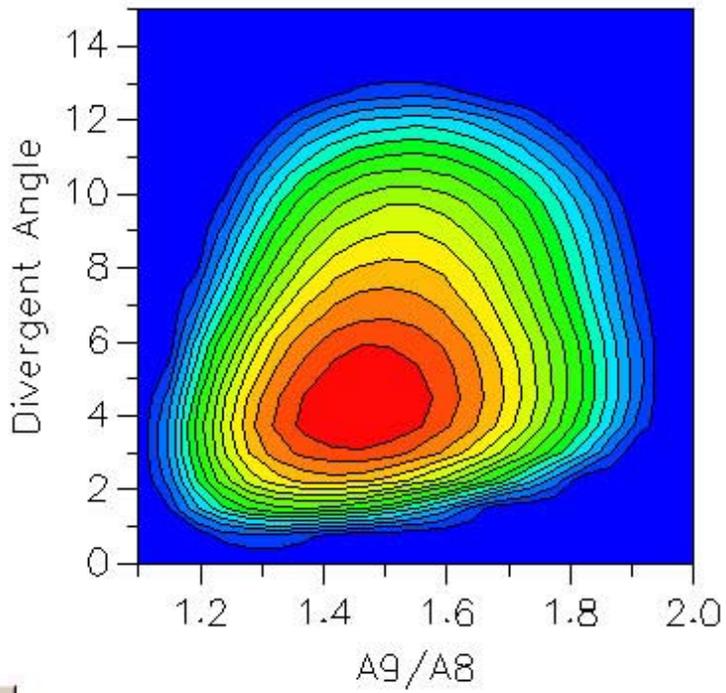
The following data window is displayed for the 2-D circular nozzle when the **Contours/Plot** button is pressed.

	A9/A8	Divergent Angle
Variable Minimum	1.10	0.00
Variable Maximum	2.00	15.00
Number of Calcs (max = 100)	20	

Plot Contour Data	
Data Minimum	0.89002
Data Maximum	0.95640
Contour Min	0.94000
Contour Max	0.95630
Increment	0.00100

After the **Calc Cfg's** button is pressed, values of the gross thrust coefficient are calculated over the input ranges of the area ratio and divergent angle. The **Plot** and **Plot+** buttons also become active. Similar to the INLET program, the **Plot** button displays a contour plot where the mouse arrow becomes a cross-hair when the user moves the mouse across the contour plot. This feature allows the user to capture and display at the bottom of the screen the coordinates and gross thrust coefficient value by pressing the left mouse button. Pressing the **Plot+** button produces an Olectra Chart contour plot of the gross thrust coefficient versus nozzle area ratio and divergent flap angle as shown on the next page.

Nozzle Thrust Coefficient Contours
Date:



C _g	
Red	0.9553 .. 0.9563
Orange	0.9543 .. 0.9553
Yellow-Orange	0.9532 .. 0.9543
Yellow	0.9522 .. 0.9532
Light Yellow	0.9512 .. 0.9522
Yellow-Green	0.9502 .. 0.9512
Light Green	0.9492 .. 0.9502
Green	0.9482 .. 0.9492
Light Green	0.9471 .. 0.9482
Green	0.9461 .. 0.9471
Cyan	0.9451 .. 0.9461
Light Blue	0.9441 .. 0.9451
Blue	0.9431 .. 0.9441
Dark Blue	0.9420 .. 0.9431
Very Dark Blue	0.9410 .. 0.9420
Black	0.9400 .. 0.9410

Print

Done

4. SKETCH Window

After design calculations have been performed for a 2-D circular convergent-divergent nozzle, pressing the **Sketch** button on the Main window opens the Sketch window and displays the outline of the nozzle at the design point as shown below for a nozzle with afterburning. The nozzle without afterburner operation is shown in black and the nozzle with afterburner operation is shown in red. The lengths and radii of the nozzle are given on the left of the sketch. The bit map image on the Sketch Pad can be copied to the Windows clip board by pressing the “Alt” and “PrtSc” keys simultaneously. From the Window’s clip board, the image can be pasted into the Paint program and saved to a file for later use.

